

National Aeronautics and Space Administration



# *Space Shuttle Stiffener Ring Foam Failure Analysis, a Non-conventional Approach.*



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# ***Space Shuttle on Ascent***



# *SRB Recovery*



*Shuttle Legacy Flight Hardware will Fly on the Space Launch System*





# *Stiffener Ring*



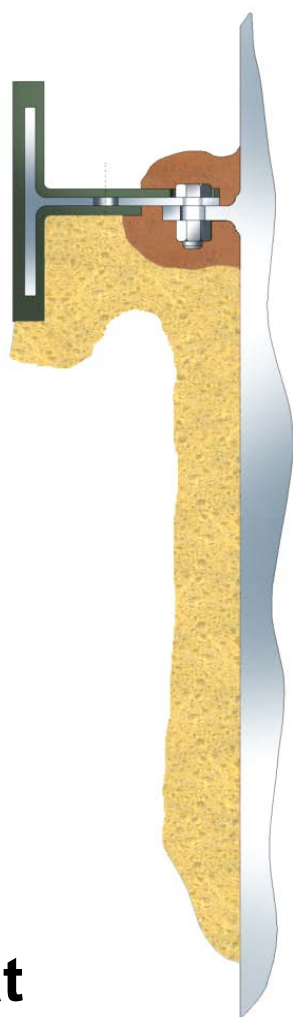
# *SRB Foam Buildup*



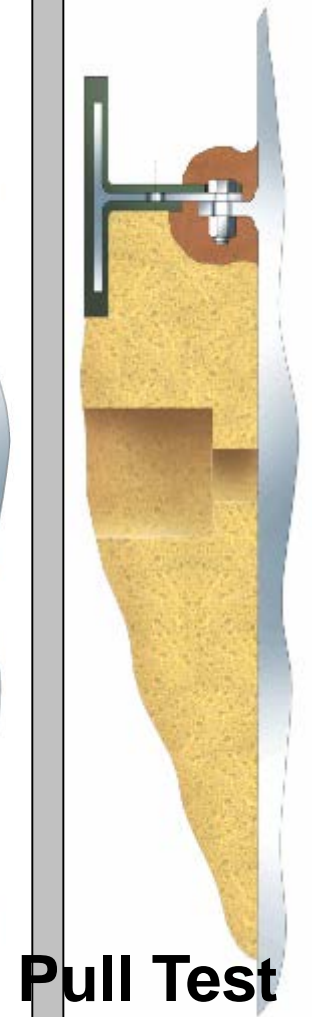
RT 455



1/8 bond coat



Coating Buildup



Pull Test

>20psi

## *Qualitative Inspection*



- ATK define the foam failures qualitatively by visual inspection of the presence or absence of foam residue on the de-bonded surface.
- Foam Failures fall into two categories
  - Adhesive
  - Cohesive
    - Mixture of both



# *Solid Rocket Booster Stiffener Ring Foam Failure*





# *Solid Rocket Booster Stiffener Ring Foam Failure*



The classical methods of analysis failed to provide a root cause into this foam failures for the 25 years.

- Chemistry-extraction of residues
- Bulk property mechanical testing
- Fracture analysis
- No known nondestructive analysis
- 10,000's of hours testing “process” variables



**A new approach was needed**

**How would a microscopist look at this?**

Cell morphology determines the mechanical strength of the foam.

Foam is the ideal media to preserve its own failure.

Cross sectioning to observe the cell morphology.

# *Foam Chemistry*



- A/B Ratio- mechanical strength and flexibility
- Blowing agent function of vapor pressure and temperature
- Exothermic reaction –driving the reaction rate
- Moisture

# *Application parameters*



- Optimum two part A/B ratio
  - Viscosities
  - Delivery Pressures
- Temperature
  - Substrate temperature
    - Ambient- outdoor conditions
  - Exothermic reaction ~140 F
- Operator application technique
  - Spray pattern
- Formulation changes
  - Blowing agent
  - Catalyst
- Humidity-dew point
  - Cure rate
  - Condensation on substrate

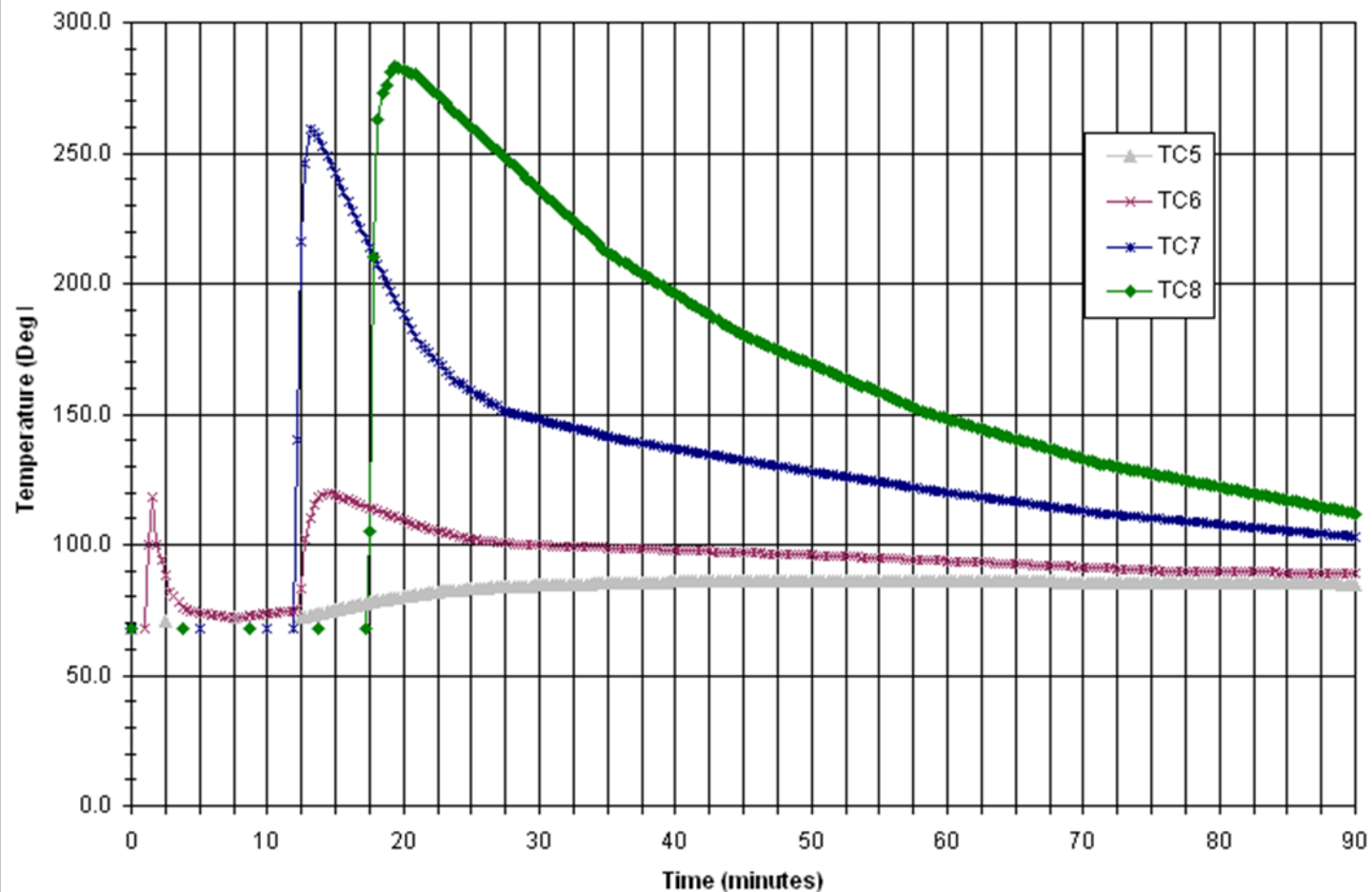




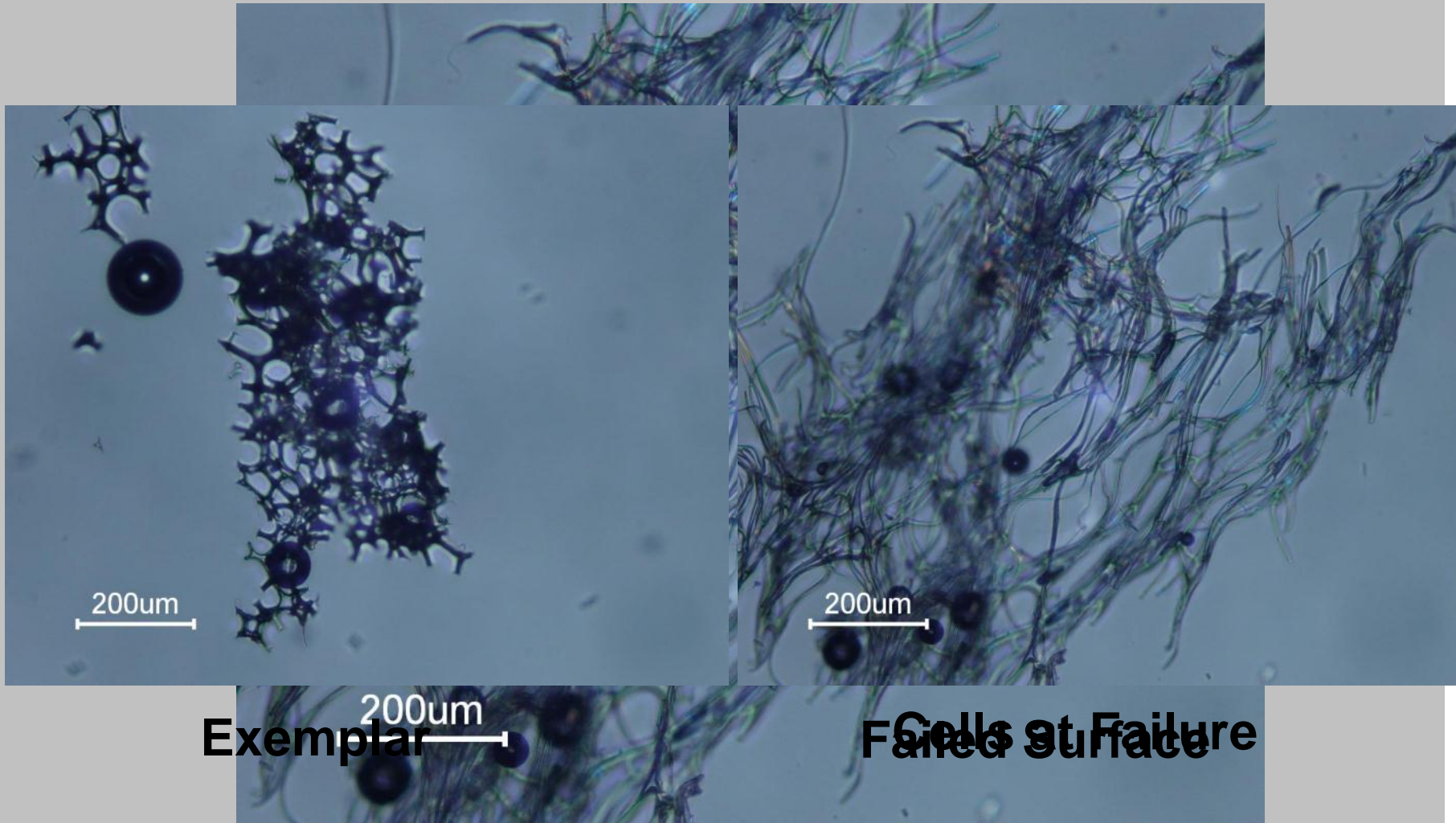
# Foam Curing $\Delta T$



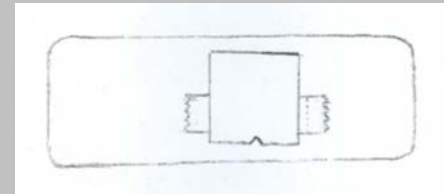
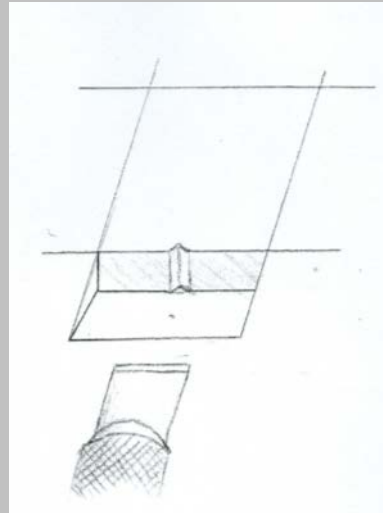
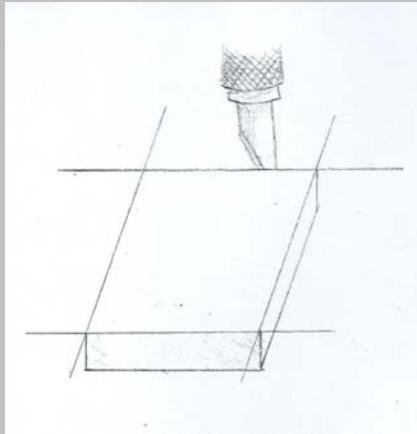
Substrate T	—	2 <sup>nd</sup> Layer T	—
Tackcoat T	—	3 <sup>rd</sup> Layer T	—



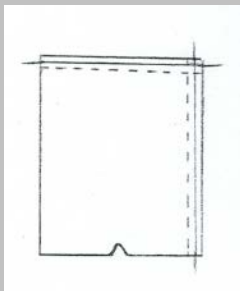
# *PLM of SRB Foam Failure*



# Cross Sectioning of Foam

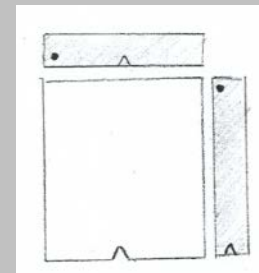


- Plastic slide with double
- sided tape

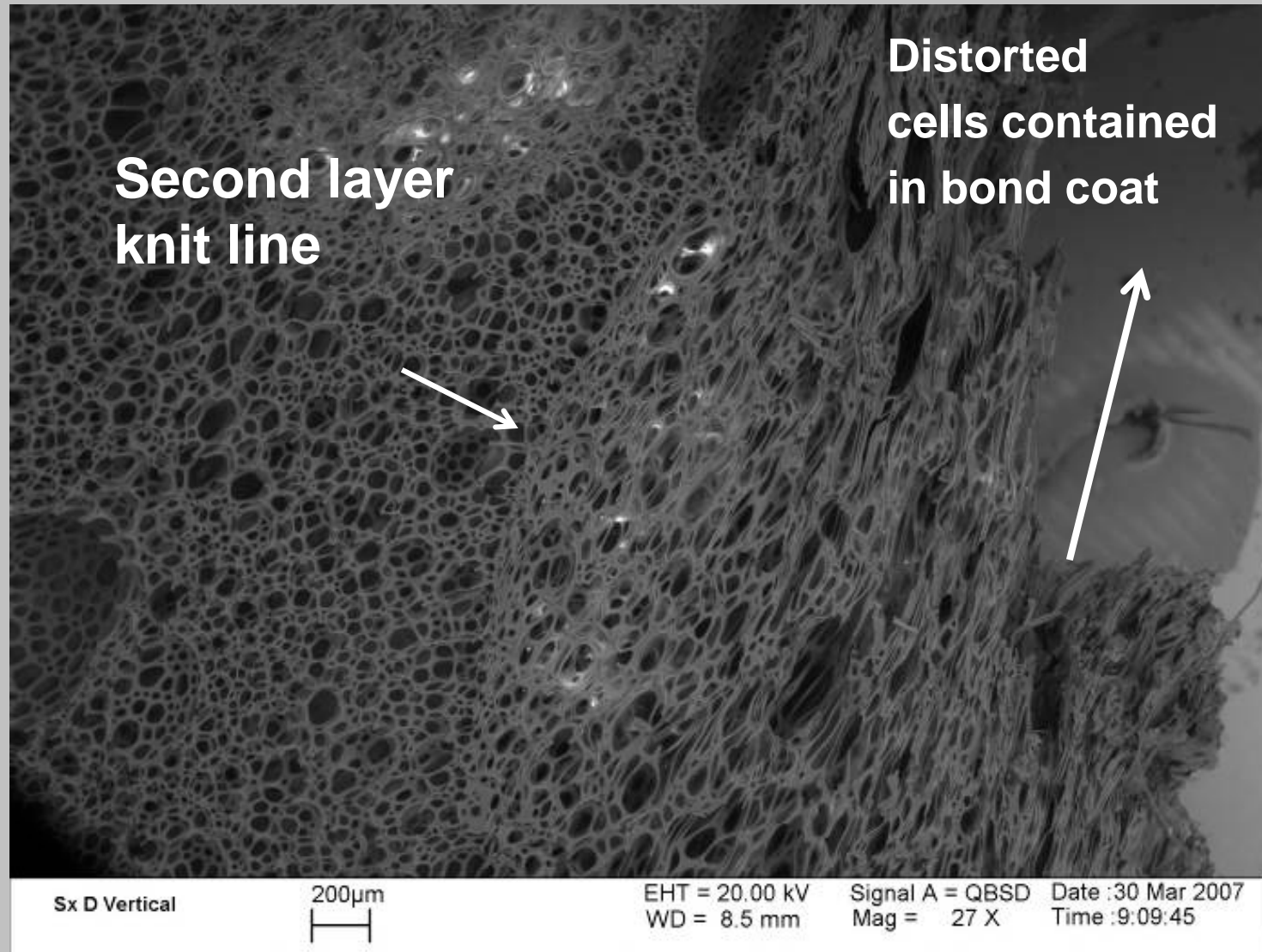


- 1<sup>st</sup> cut 1 mm
- section single edge
- razor

- 2<sup>nd</sup> cut 0.5-1mm
- section double
- edge razor



# *Cross Section of SRB Foam Failure*



**Chief SRB Engineer—we have never looked at foam like this**

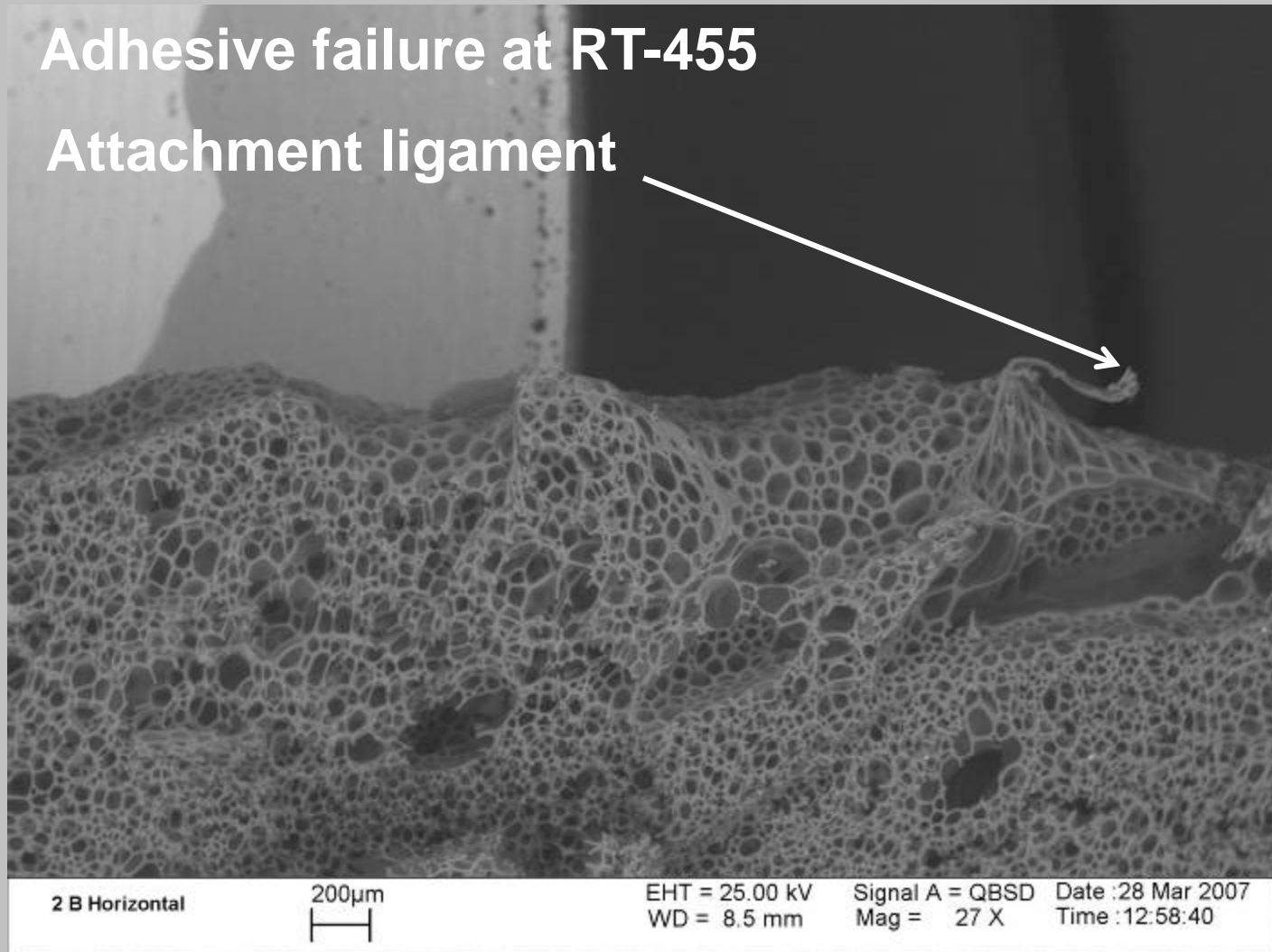


# *SRB Foam Failure*

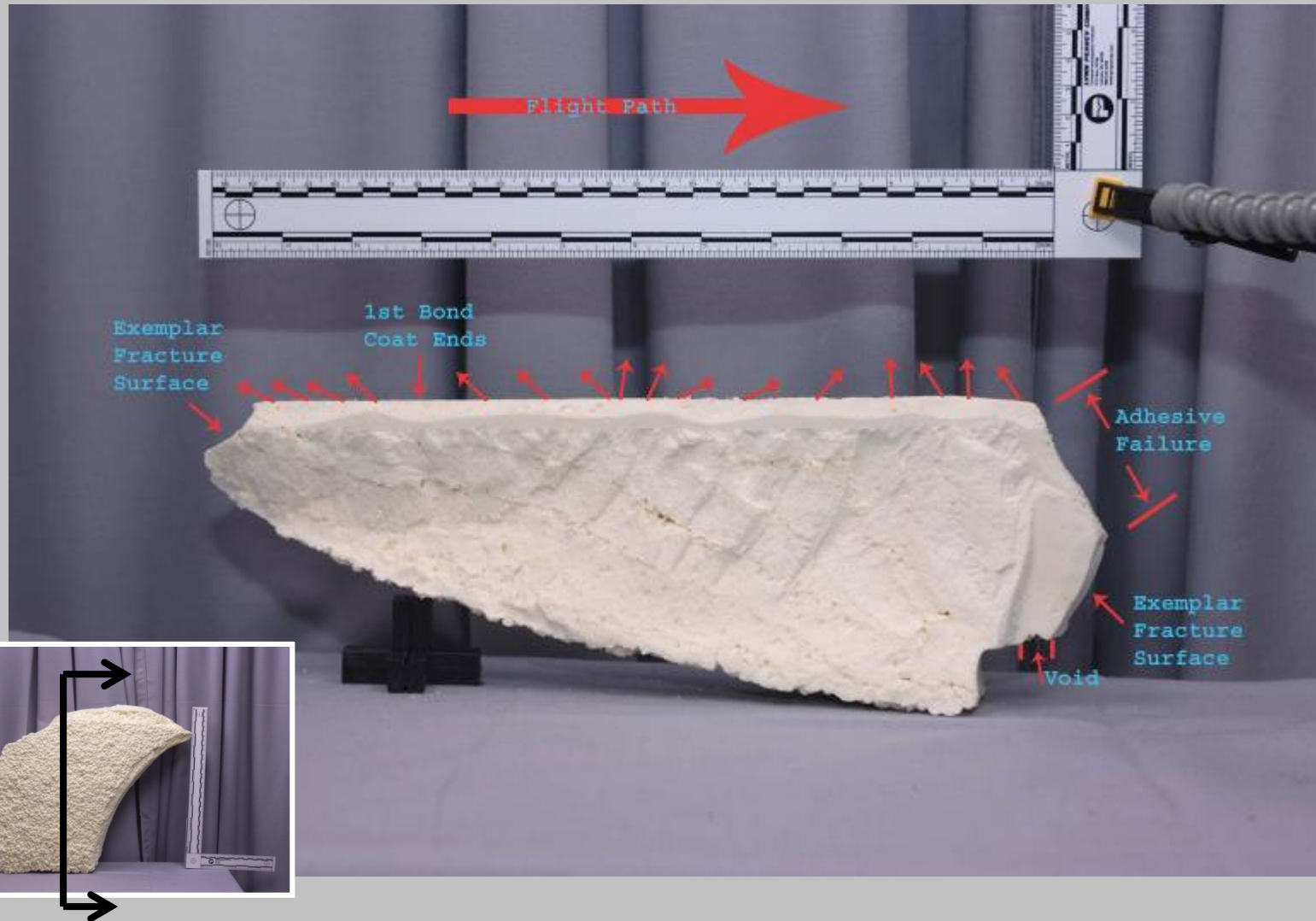


**Adhesive failure at RT-455**

**Attachment ligament**



# *SRB Foam Failure*



# *SRB Foam Failure*



>80 percent application failed

The foam bond coat displayed two modes of failure:

- Cohesive failure was observed due to severely deformed foam cells in the bond coat
- Adhesive failure was observed at the RT-455 Epoxy interface.

The observed morphology indicates that the bond coat was not fully cured before other forces were applied, e.g. the expansion forces of the second coat distorted the bonding cells.



## *Process Changes*

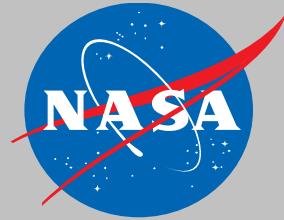
- Substrate as an infinite heat sink
- Develop spray hardware to apply micro bond layers  
     $\ll 1/8''$  layers
- Primer (polyurethane)
- The bond coat should be allowed to completely dry/cure
  - Never apply over wet uncured layers



# *Sunset on International Space Station Expedition 15*



ISS015E10469

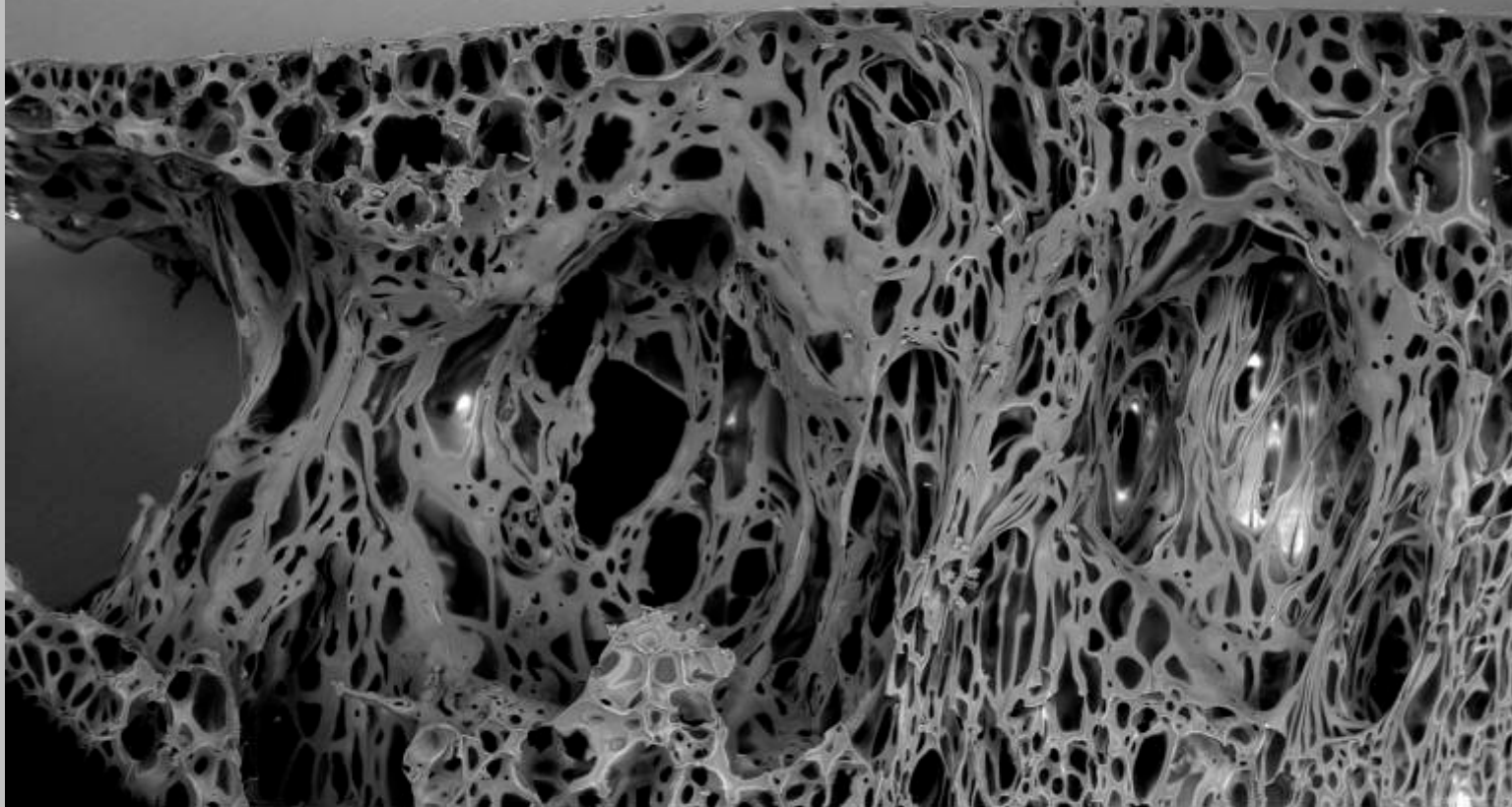


# *A View from Above!*

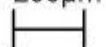


ISS007E07306

•30 % adhesive failure



MSL-2007-0134  
# 1

200µm  


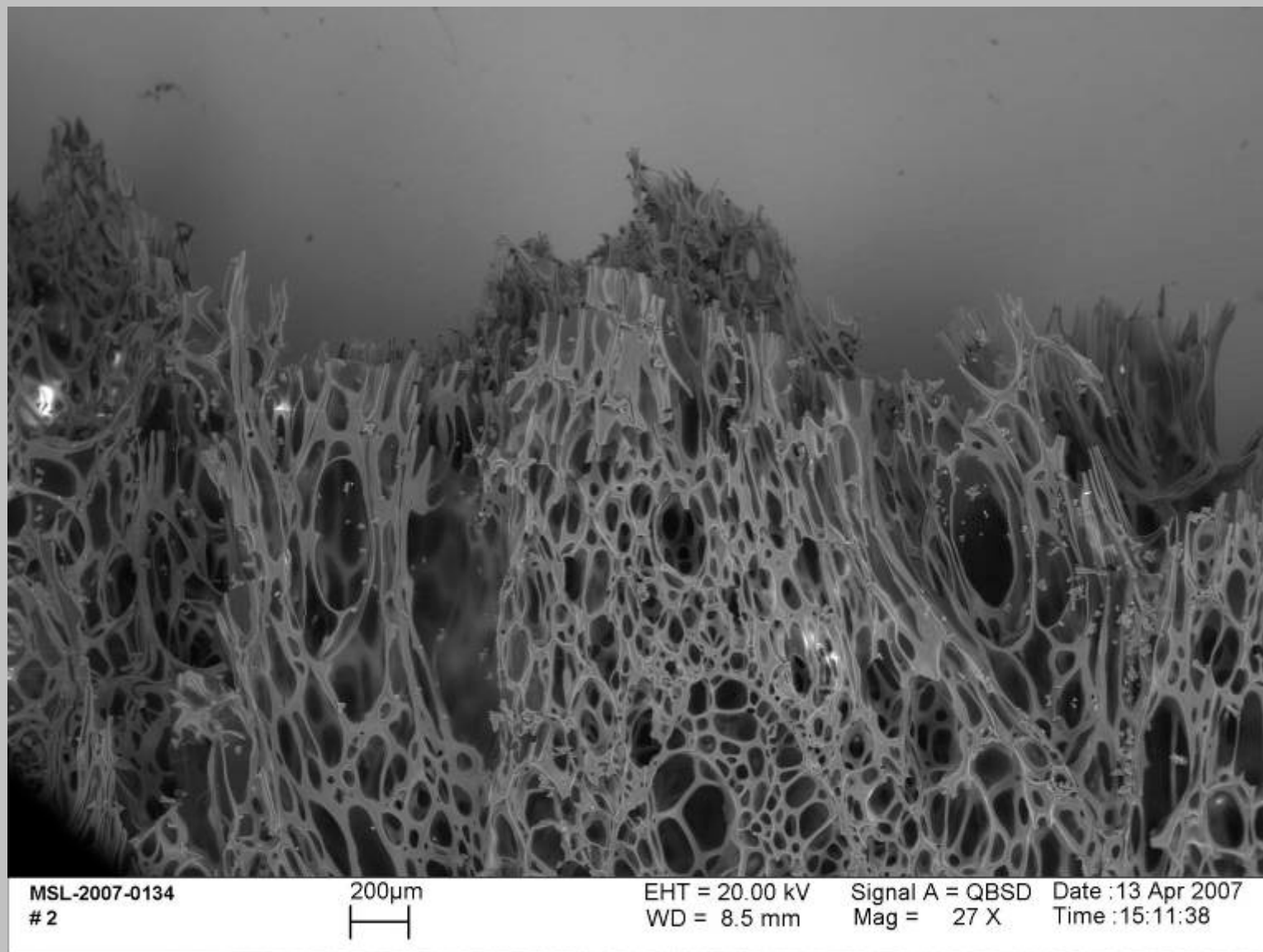
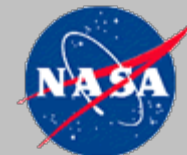
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WD = 8.5 mm

Signal A = QBSD  
Mag = 27 X

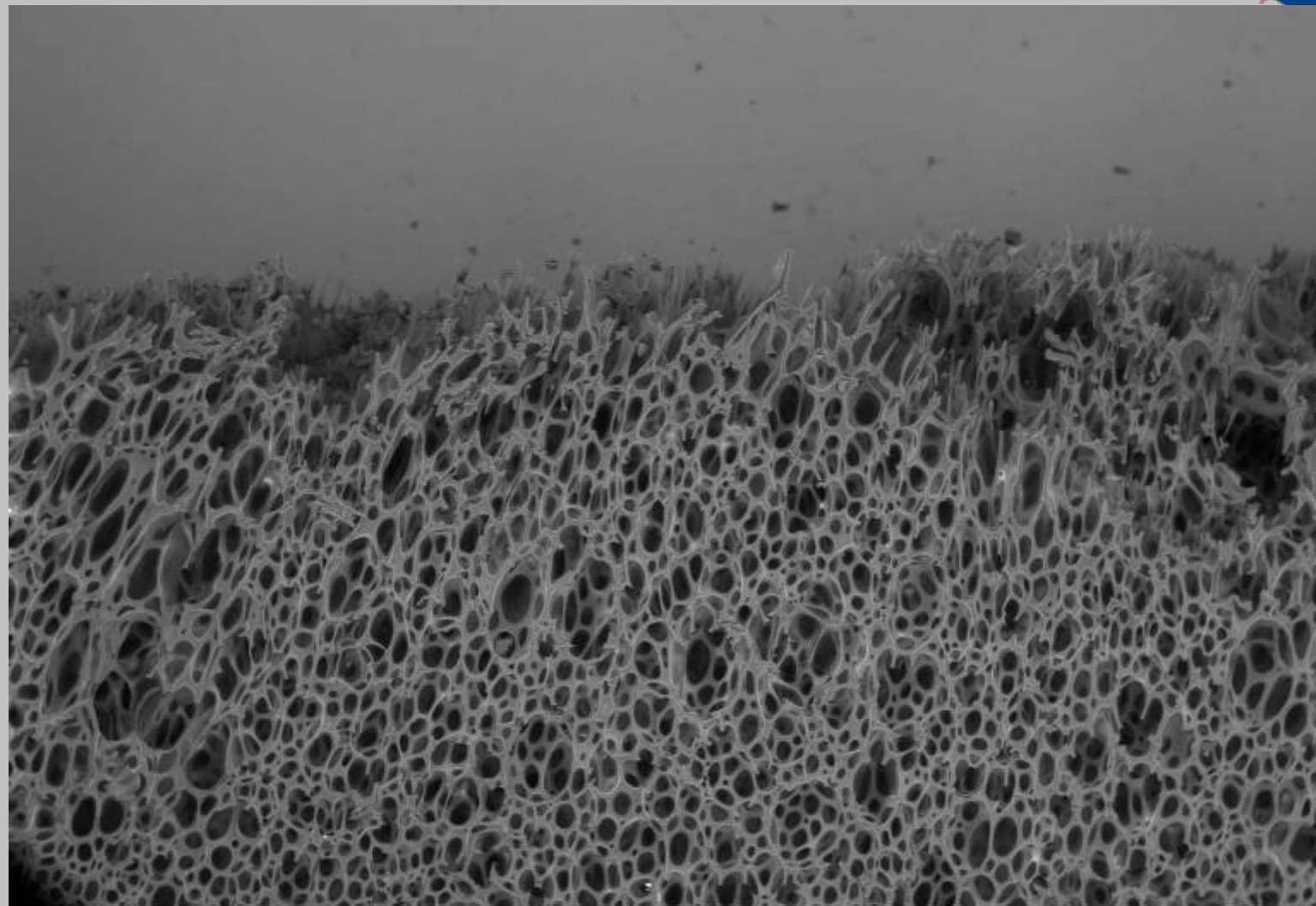
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Time : 14:46:38



50 PSI



78 PSI



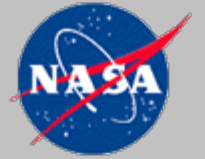
MSL-2007-0134  
# 3

200µm  
—|—|

EHT = 20.00 kV  
WD = 7.5 mm

Signal A = QBSD  
Mag = 30 X

Date : 13 Apr 2007  
Time : 15:27:54



# *Foam Mechanics*



- Mixing of part A and B
- Evaporation of blowing agent-bubble formation
- Catalyst initiate cure
- Exothermic reaction